

AE31002 Aerospace Structural Dynamics

Damped SDOF System

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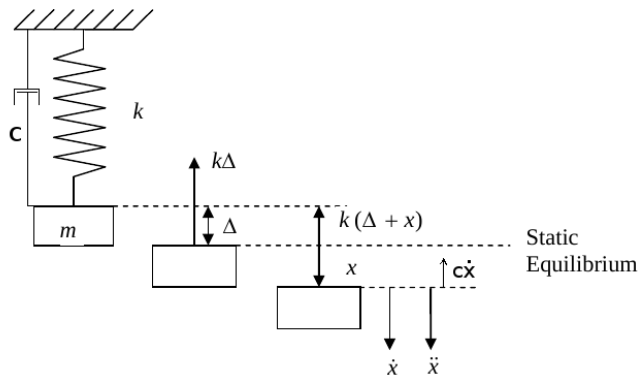
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Viscous Damping – the type of damping force that could be developed in a body restrained in its motion by a surrounding viscous fluid.



The summation of forces following
D'Alembert's principle

$$m\ddot{x} + c\dot{x} + kx = 0$$

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The trial solutions $x = A \sin \omega t$ or $x = B \cos \omega t$ does not satisfy the above equation where as $x = Ae^{\lambda t}$ satisfies.

$$\Rightarrow m\lambda^2 + c\lambda + k = 0$$

$$\Rightarrow \lambda^2 + \frac{c}{m}\lambda + \frac{k}{m} = 0$$

$$\lambda = -\left(\frac{c}{2m}\right) \pm \left[\left(\frac{c}{2m}\right)^2 - \left(\frac{k}{m}\right)\right]^{\frac{1}{2}}$$

$$x = e^{-\left(\frac{c}{2m}\right)t} \left[Ae^{\sqrt{\left(\frac{c}{2m}\right)^2 - \left(\frac{k}{m}\right)} t} + Be^{-\sqrt{\left(\frac{c}{2m}\right)^2 - \left(\frac{k}{m}\right)} t} \right]$$

There may be three possible cases where the portion under the radical may be positive or negative or zero for the cases over-damped or under-damped or critically-damped cases, respectively.

Case I $\left(\frac{c}{2m}\right)^2 > \left(\frac{k}{m}\right)$ – **Over Damped Case**

The solution becomes $x = A_1 e^{-\lambda_1 t} + A_2 e^{-\lambda_2 t}$ where λ_1 and λ_2 are two real roots of the equation. The system displacement will decay without any oscillation. This equation shows that there is a usual creeping back of the mass to the equilibrium position. This is a case of over-damped situation and no oscillation is possible.

ANIMATION OVER DAMPED**Case II** $\left(\frac{c}{2m}\right)^2 < \left(\frac{k}{m}\right)$ – **Under Damped Case**

Roots of the equation are imaginary.

Oscillation continues for a considerable time period depending upon the damping present in the system.

$$\lambda = -\left(\frac{c}{2m}\right) \pm i \left[\left(\frac{k}{m}\right) - \left(\frac{c}{2m}\right)^2 \right]^{\frac{1}{2}}$$

Now using exponential series,

$$e^{i\theta} = \cos\theta + i \sin\theta$$

$$e^{-i\theta} = \cos\theta - i \sin\theta$$

Substituting the expressions similar to the above we get,

considering, $\omega_d = \left[\left(\frac{k}{m} \right) - \left(\frac{c}{2m} \right)^2 \right]^{\frac{1}{2}}$

$$\begin{aligned} x &= e^{-\frac{c}{2m}t} \left[A_1' e^{i\omega_d t} + A_2' e^{-i\omega_d t} \right] \\ &= e^{-\frac{c}{2m}t} \left[A_1' (\cos \omega_d t + i \sin \omega_d t) + A_2' (\cos \omega_d t - i \sin \omega_d t) \right] \\ &= e^{-\frac{c}{2m}t} \left[(A_1' + A_2') \cos \omega_d t + i(A_1' - A_2') \sin \omega_d t \right] \end{aligned}$$

Let,

$$A'_1 + A'_2 = A \sin \phi$$

$$i(A'_1 - A'_2) = A \cos \phi$$

$$x = e^{(-\frac{c}{2m}t)} [A \sin(\omega_d t + \phi)]$$

where,

$$\begin{aligned} A^2 &= (A'_1 + A'_2)^2 - (A'_1 - A'_2)^2 \\ &= A_1'^2 + A_2'^2 + 2A'_1 A'_2 - A_1'^2 - A_2'^2 + 2A'_1 A'_2 \\ &= 4A'_1 A'_2 \end{aligned}$$

$$A = 2(A'_1 A'_2)^{1/2}$$

$$\tan \phi = \frac{A'_1 + A'_2}{i(A'_1 - A'_2)}$$

So, roots of the equation are,

$$\lambda_1 = -\frac{c}{2m} + i\omega_d$$

$$\lambda_2 = -\frac{c}{2m} - i\omega_d$$

With a frequency, $\omega_1 = \omega_d = \left[\left(\frac{k}{m} \right) - \left(\frac{c}{2m} \right)^2 \right]^{\frac{1}{2}}$

The frequency is not same as of the case of undamped case. It is some amount less than the previous value, i.e., $\sqrt{\frac{k}{m}}$. The change is dependent on the damping parameter c and the mass of the system.

Therefore the equation of damped single degree of freedom system is

$$x = e^{\left(-\frac{c}{2m}t\right)} [A \sin(\omega_d t + \phi)]$$

ANIMATION UNDER DAMPED

Case III $\left(\frac{c}{2m}\right)^2 = \left(\frac{k}{m}\right)$ – **Critically Damped Case**

If c_c is the coefficient for critical damping.

$$c_c^2 = 4km \Rightarrow c_c = 2\sqrt{km}$$

$$\omega_n = \sqrt{\frac{k}{m}} \Rightarrow c_c = 2m\omega_n \Rightarrow c_c = \frac{2k}{\omega_n}$$

Damping ratio $\xi = \frac{c}{c_c} = \frac{\text{Actual Damping Coefficient}}{\text{Critical Damping Coefficient}}$

$$\frac{c}{2m} = \frac{\xi c_c}{2m} \text{ substituting the above relation}$$

$$\frac{c}{2m} = \xi \omega_n$$

The equation for free damped single degree freedom system modifies to

$$x = e^{(-\xi\omega_n t)} [A \sin(\omega_d t + \phi)]$$

$$\text{where } \omega_d = \omega_n \sqrt{1 - \xi^2}$$

$$T_1 = \frac{2\pi}{\omega_d} = \frac{2\pi}{\omega_n \sqrt{1-\xi^2}},$$

$$\lambda_{1,2} = -\xi\omega_n \pm \sqrt{(\xi\omega_n)^2 - \omega_n^2} = \left(-\xi \pm \sqrt{\xi^2 - 1}\right) \omega_n$$

This is the case of $\xi < 1$, i.e., All damped free vibrations.

when $\xi = 1$ It is a case of critically damped one.

$\xi > 1$ It is a case of over-damped one.

In case of under-damped or damped case the constants may be found out solving a initial value problem.

Let, at $t = 0$, $x = x_0$, and $\dot{x} = v_0$ and if we substitute it to
 $x = e^{(-\xi\omega_n t)} [A \sin(\omega_d t + \phi)]$

We get

$$x = e^{(-\xi\omega_n t)} \left[x_0 \cos \omega_d t + \frac{v_0 + x_0 \xi \omega_n}{\omega_d} \sin \omega_d t \right]$$

$$\Rightarrow x(t) = C e^{(-\xi\omega_n t)} \cos(\omega_d t - \alpha)$$

where $C = \sqrt{x_0^2 + \left(\frac{v_0 + x_0 \xi \omega_n}{\omega_d} \right)^2}$

and $\tan \alpha = \frac{v_0 + x_0 \xi \omega_n}{\omega_d x_0}$

ANIMATION 2

In case of Over-Damped System, we get from the general solution

$$x = Ae^{(-\xi + \sqrt{\xi^2 - 1})\omega_n t} + Be^{(-\xi - \sqrt{\xi^2 - 1})\omega_n t}$$

where $A = \frac{\dot{x}_0 + (\xi + \sqrt{\xi^2 - 1})\omega_n x_0}{2\omega_n \sqrt{\xi^2 - 1}}$ and $B = \frac{-\dot{x}_0 - (\xi - \sqrt{\xi^2 - 1})\omega_n x_0}{2\omega_n \sqrt{\xi^2 - 1}}$

ANIMATION OVER DAMPED

In case of Critically-Damped System, we get from the general solution

$$x = (A + Bt)e^{\omega_n t}$$

With implementation of initial conditions

$$x(t) = [x_0 + \{\dot{x}_0 + \omega_n x_0\}t] e^{\omega_n t}$$

ANIMATION of CRITICALLY DAMPED SYSTEM